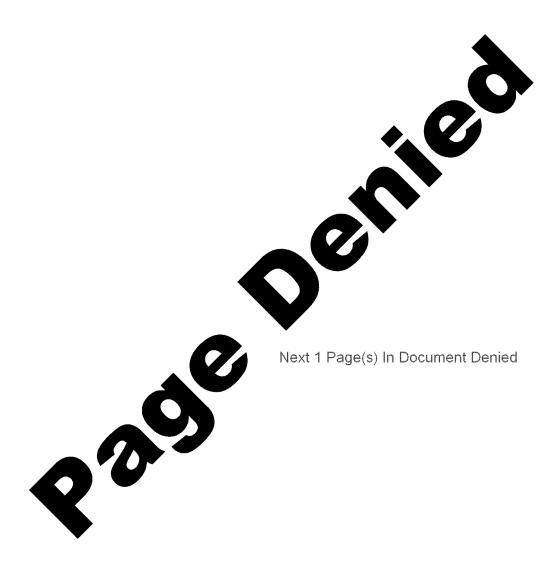
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TOP SEGNAL

PROJECT CORONA

COR 0422 \* Copy 1 of 2

Army review(s) completed.

TOP SECRET

#### INDEX

Operational Reports Control - Tab A

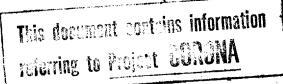
lst and 2nd CORONA Launch - Tab B

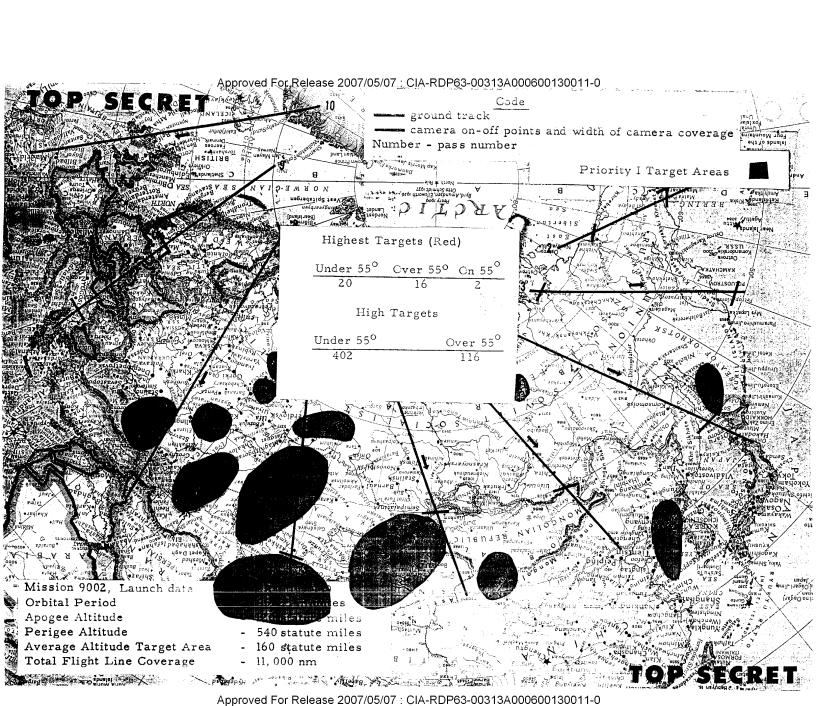
Sequence of Events,
CORONA Test Launch #3
Chronological Sequence of - Tab C
Events CORONA Mission
9001 - June 1959

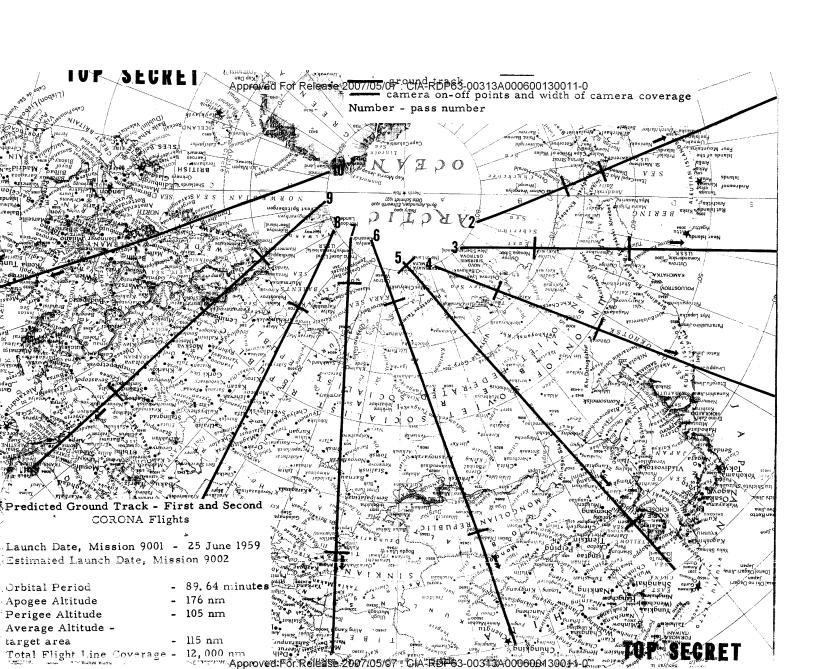
Time of Year Limitations - Tab D

# Approved For Release 2007/05/07: CIA-RDP63-00313A000600130011-0 Operational Reports Control

Time	Report	Title	Reason for Report
Each Friday			To keep Hqs informed on the long range status (estimated launch dates).
L-14 days		Ephemeris Data	To provide Hqs with ground track and othed data necessary to establish camera on-of
L-10 days		Program Data	points. To furnish LMSD with data to cut prograntage (camera on-off points). Data may befurnished for more than one tape.
L-5 days		Daily Status	To keep Hqs informed on the immediate capability to launch. Report required eved day from L-5 days until launch.
L-3 days		Mission Data	To inform LMSD of camera settings and type to be used. Also to select final cam on-off points to use.
L-30 hours		Mission Plan	To inform LMSD of Hqs decision to delay continue the launch based on Weather in t get area. To provide LMSD with optimum launch time and earliest and latest time of day launch could occur and give sufficient
L-8 hours		Recovery Wx	sunlight for photography.  To provide Hqs with forecast of Wx in Recovery area
L-6 hours		Go-No-Go	To inform LMSD of final Hqs decision to launch or delay based on target weather.
L-2 hours	1	Special Status	To keep Hqs informed of status of vehicle during pre-launch countdown. Report su mitted each hour until launch or abort.
L110 min		Launch Report	To inform Hqs that launch has occurred.
L12 hrs		Progress Report	To inform Hqs of success of launch and preliminary data.
L420 hrs		Progress Report	To provide Hqs with actual ground track satellite and estimated recovery time.
ASAP		Flash Recovery	To inform Hqs of the recovery of the cap sule and brief description of condition.
ASAP		Take Report	To inform Hqs of ETA of take at West Co and condition of take.
As Occurs		Abort Report	To inform Hqs of a delay requiring post- ponement of launch until next day or long
As Occurs		Crash Report	To inform Hqs of a mishap to a vehicle i volving loss of life, injury or destruction







## Approved For Release 2007/05/07 : CIA-RDP63-00313A000600130011-0

# Chronological Sequence of Events - CORONA Test Vehicle #3 April 1959

Date/Time	Event
13/1800Z	Estimated time of launch of test vehicle #3.
l April L-12 days	Receip by Headquarters of Ephemeris Data (from this ground track of the vehicle is obtained), recommended camera settings and film type.
2 April L-11 days	Headquarters determined camera on and off points considering intelligence requirements, film supply, and climatology.
8/2100Z L-5 days	Headquarters received first of daily status reports on vehicle with latest estimated time of launch (ETL).
10/2000Z L-72 hours	Headquarters considered and approved recommended camera settings and film type.
12/1200Z L-30 hours	Headquarters considered weather in target area (best weather to date - 4000 nm flight lines in 3/8 or less cloud cover). Headquarters also determined earliest and latest time of day launch could occur and give sufficient sunlight for photography.
13/1200Z L-6 hours	Headquarters took final look at weather (2500 nm flight lines in 3/8 or less cloud cover).
13/1600Z L-2 hours	Receipt by Headquarters of first of hourly status reports giving latest ETL and reasons for delays.
13/1700-2100Z	Receipt by Headquarters of several launch delays caused primarily by weather at the launch site.
13/2200Z L142 minutes	Receipt of launch report. Successful launch at 2118Z. Vehicle definitely in orbit.
13/2318Z L42 hours	Receipt by Headquarters of first post launch status report.  Vehicle in polar orbit. Orbital time approximately 3 minutes less than planned. Vehicle under control.

14/1818Z L**12**1 hours Receipt of special status report indicating command control of vehicle has been lost. Recovery operation cancelled due to capsule impact predicted in area 500 miles north of Spitzberg Islands at 15/0018Z.

14/2118Z L424 hours

Receipt of second status report. Orbital time 90.5 minutes vs. 94.5 minutes predicted. Actual ephemeris indicates each pass falls successively further to the East of predicted path.

15/2000Z L447 hours Receipt of special status report. The VAFB tracked vehicle and received telemetry on pass 24 and 25. Good data with positive indication of capsule separation. On the 26th pass, no telemetry received indicating batteries dead. Also unconfirmed report of flash in sky followed by report of parachute descending in Spitzberg Island area. Air Force investigating.

16 April

Brief post mortem. Launch was successful with vehicle going into polar orbit of shorter duration than anticipated. Vehicle was stabilized, command control initially established and the capsule did separate. Separation occurred at wrong time because the programmer could not be changed from predicted to the actual or bit time. Programmer was over-controlled by Hawaiian tracking station in an attempt to adjust to difference in actual vs. predicted orbital time. An automatic feature of the programmer causes it to slip back to the predicted time when so over-controlled. Since programmer is set to receive commands only during certain times, command control was not achieved again because when the vehicle was in range of our ground stations, the programmer would not accept the signals. (Results of programmer operating on predicted, not actual, time.) Based on the predicted time the programmer automatically orders the recovery process to commence resulting in capsule release in the wrong area. A review of actual weather conditions over the target area during orbital time indicated 3000 nm of flight lines with 3/8 or less cloud cover. From a photo reconnaissance standpoint, correct programmer timing being achieved and recovery accomplished, this would have been considered a successful mission.

### Chronological Sequence of Events CORONA Mission 9001 - June 59

	Date/Time	Event
	20 May L- 33 days	Receipt by Headquarters of ephemeris date (Ground Track)
	22 May L- 35 days	Selection of film type and camera setting, camera on off points, and launch time limits by Headquarters. Info dispatched to Palo Alto.
	19 June L- 5 days	Headquarters received 1st Daily Status with ETL of 23/1800Z (launch set for earliest time authorized by Headquarters.)
	22/1500Z L- 30 hours	Headquarters considered target weather (3800NM of flight lines in 3/8 or less cloud cover.)
	23/1200Z L- 6 hours	Headquarters took final look at weather (3200NM of flight lines in 3/8 or less with good possibility of small areas opening in Urals and Western Russia) Approval Sent.
·	23/1500Z	Back up weather (for launch on 24th). Considered - Weather had deteriated.
25X1	23/1700Z	Receipt of Status Report) - Holding at L minus 95 minutes because range safety line out - Vehicle condition excellent.
25X1	23/1800 to 1900Z	Receipt of several indicating still holding for range safety lines.
25X1	23/1950Z	Receipt of (Abort Report). Launch rescheduled to 25/1800Z. Reason given was range safety lines out and 48 hours required for repair - necessary rescheduling and notification at Headquarters was performed.
	24/1500Z	Considered target weather (2000NM in 3/8 or less).
	24/2100Z	Daily Status Report received reconfirming ETL on 25/1800Z.

- 2 -

#### Continued

Date/Time	Event
25/1200Z	Considered final weather - some improvement over forecast made on 24th - 2400 NM in 3/8 or less - probability of additional small areas in Central and North Central USSR.
25/1700Z	Receipt of indicating now ETL of 25/1900Z because of minor technical delays - Launch prospects excellent.
25/1800Z	Series of holds caused by small technical prob- lems, pressurization, fueling problems and beacon problems.
25/2250Z	Launch occurred with initial data indicating a successful launch.
26/0110Z L plus 2:20	lst Progress Report indicated vehicle probably did not go into orbit.
26/0240Z 1 plus 4:00	2nd Progress Report indicated definite failure of vehicle to go into orbit.
26 June	Brief post mortem - Exact reason for failure to orbit unknown at this time. Departure azimuth 4 degrees west of planned which would decrease boost given by rotation of earth. THOR trajectory was lower than nominal. Insertion velocity was 25,605 Ft/sec. vs 25,691 Ft planned. Actual weather in target area would have permitted 3500NM of photography.

14/1818Z L421 hours Receipt of special status report indicating command control of vehicle has been lost. Recovery operation cancelled due to capsule impact predicted in area 500 miles north of Spitzberg Islands at 15/0018Z.

14/2118Z L124 hours

Receipt of second status report. Orbital time 90.5 minutes vs. 94.5 minutes predicted. Actual ephemeris indicates each pass falls successively further to the East of predicted path.

15/2000Z L<del>1</del>47 hours

Receipt of special status report. The VAFB tracked vehicle and received telemetry on pass 24 and 25. Good data with positive indication of capsule separation. On the 26th pass, no telemetry received indicating batteries dead. Also unconfirmed report of flash in sky followed by report of parachute descending in Spitzberg Island area. Air Force investigating.

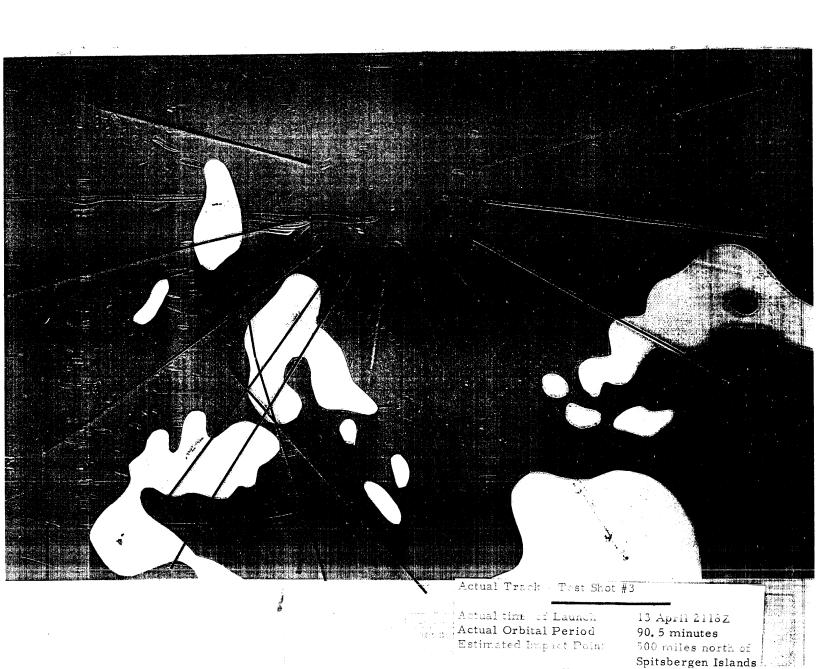
16 April

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- 2 -

#### Continued

Date/Time	Event	
25/1200Z	Considered final weather - some improvement over forecast made on 24th - 2400 NM in 3/8 or less - probability of additional small areas in Central and North Central USSR.	-
25/1700Z	Receipt of indicating now ETL of 25/1900Z because of minor technical delays - Launch prospects excellent.	25X1
25/1800Z	Series of holds caused by small technical prob- lems, pressurization, fueling problems and beacon problems.	
<b>2</b> 5/ <b>22</b> 50Z	Launch occurred with initial data indicating a successful launch.	
26/0110Z L plus 2:20	lst Progress Report indicated vehicle probably did not go into orbit.	
26/0240Z 1 plus 4:00	2nd Progress Report indicated definite failure of vehicle to go into orbit.	
26 June	Brief post mortem - Exact reason for failure to orbit unknown at this time. Departure azimuth 4 degrees west of planned which would decrease boost given by rotation of earth. THOR trajectory was lower than nominal. Insertion velocity was 25,605 Ft/sec. vs 25,691 Ft planned. Actual weather in target area would have permitted 3500NM of photography.	



Approved For Release 2007/05/07 : CIA-RDP63-00313A000600130011-0





## Approved For Release 2007/09/07 504 20 P63-00313A000600130011-0

## Time of Year Limitations on Project CORONA

#### Factors

- 1. Vehicle in one day polar orbit.
- 2. USSR and European satellites considered only areas justifying CORONA coverage.
  - 3. Concentration of priority areas between 50° and 60° N latitude.
- 4. Total vehicle linear flight line coverage in desired area equals 13,000 nm.
  - 5. Film supply will provide for 12,000 linear miles of flight line.
  - 6. Minimum acceptable sun altitude is 15°.
- 7. Weather variable but will reduce nm of flight lines open to photography by approximately 50%.

## Limitations Imposed by Ahove Factors (except weather)

- 1. 3 April to 10 September no limitations.
- 2. On 6 October and on 8 March no photography above 70° N latitude 9,000 nm flight lines available.
- 3. On 3 November and on 8 February no photography above 60° N latitude 5,000 nm flight lines available.
- 4. On 22 November and 22 January no photography above  $55^{\rm o}$  N latitude 3,600 nm flight lines available.

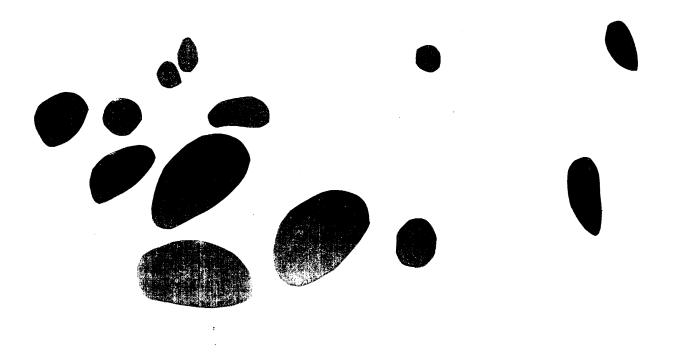
### Recommendations

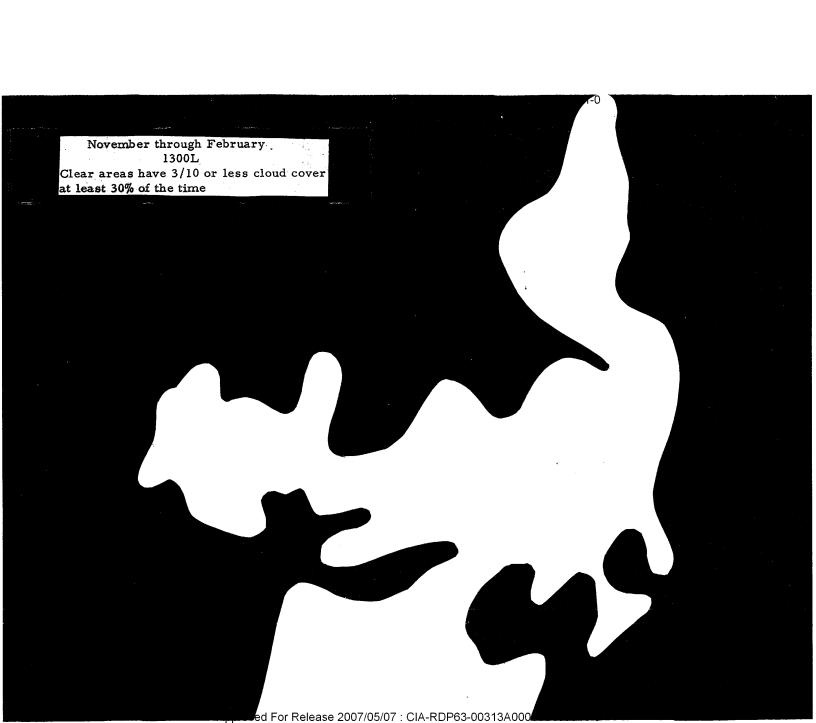
- 1. The present schedule should be either contracted or expanded to preclude CORONA photo launches within period early November to mid-February.
  - 2. Shift target areas to South, i.e., China, Tibet, Mongolia, Indo China.
- 3. Vary launch azimuth (inclination angle) to allow longer passes over USSR below 60 N latitude.
  - 4. Use an alternate payload of interest to the Intelligence Community.
- 5. Use an alternate payload of scientific interest and useful in advancing our cover story.

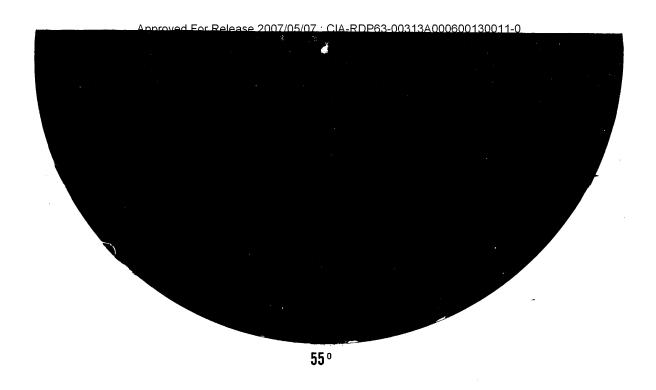
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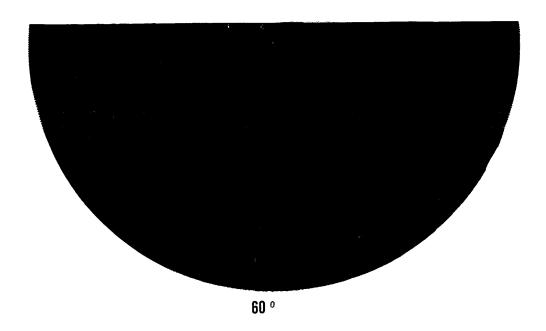
Approved For Release 2007/05/07: CIA-RDP63-00313A000600130011-0

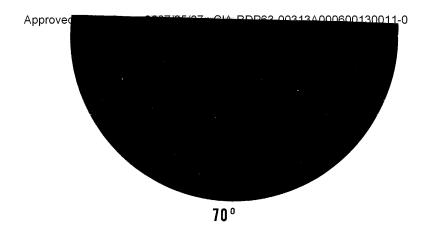
Priority I Target Areas

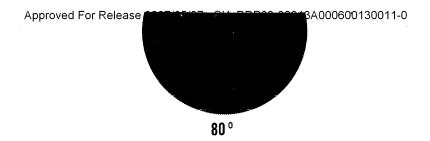






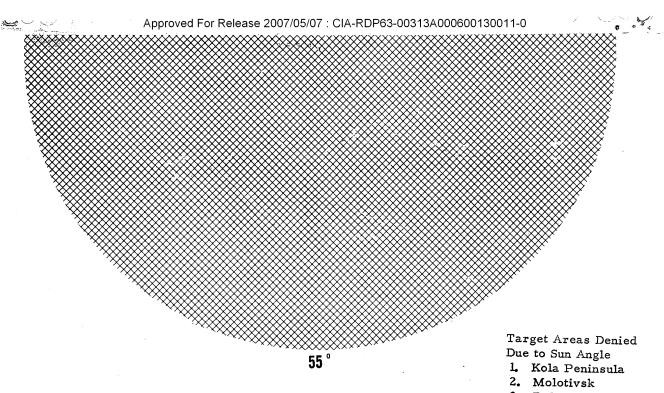






Approved For Release 2007/05/07: CIA-RDP63-00313A000600130011-0

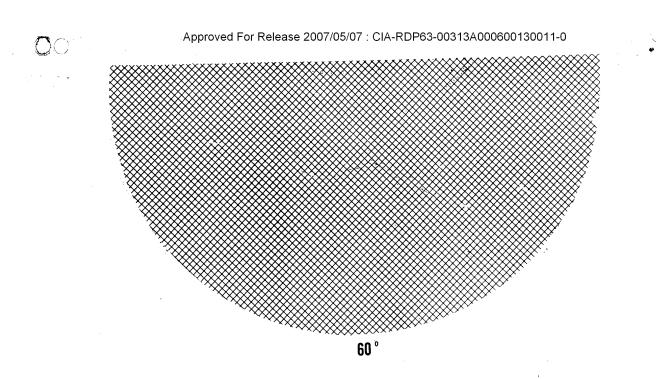
Priority I Target Areas



- 3. Polyarnyy Urals
- 4. N. Portion of Urals
- 5. Moscow Complex
- 6. Kazan and Engels
- 7. Portion of Kuznets Basin

 $15^{\,\rm o}$  sun angle

Area with sun below 15° on 22 Nov



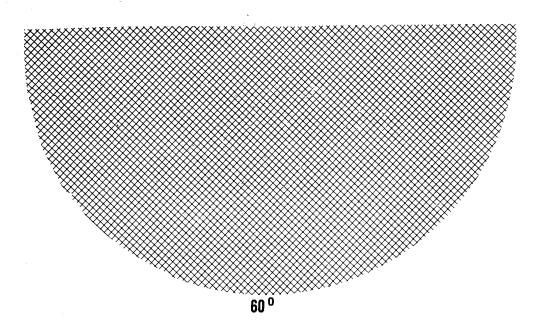
Target Areas Denied Due to Sun Angle

- 1. Kola Peninsula
- 2. Molotovsk
- 3. Polyarnyy Urals
- 4. Northern tip of Urals

15° sun angle

Area with sun below 15° on 26 Oct

Approved For Release 2007/05/07 : CIA-RDP63-00313A000600130011-0



Target Areas Denied Due to Sun Angles

- l. Kola Peninsula
- 2. Molotovsk
- 3. Polyarnyy Urals4. Northern tip of Urals

 $10^{\rm o}$  sun angle

Area with sun below 10° on 12 Nov

